

# The helicopter (Skeldar) blade dynamics

## General rigid body coordinate transformations

$$x_E = x_{B,0} + A_{B2E} x_B$$

$$x_B = x_{h,0} + A_{h2B} x_h$$

$$x_h = x_{b,0} + A_{b2h} x_b$$

$$\dot{x}_E = \dot{x}_{B,0} + A_{B2E} (\omega_B \times x_B) + A_{B2E} \dot{x}_B$$

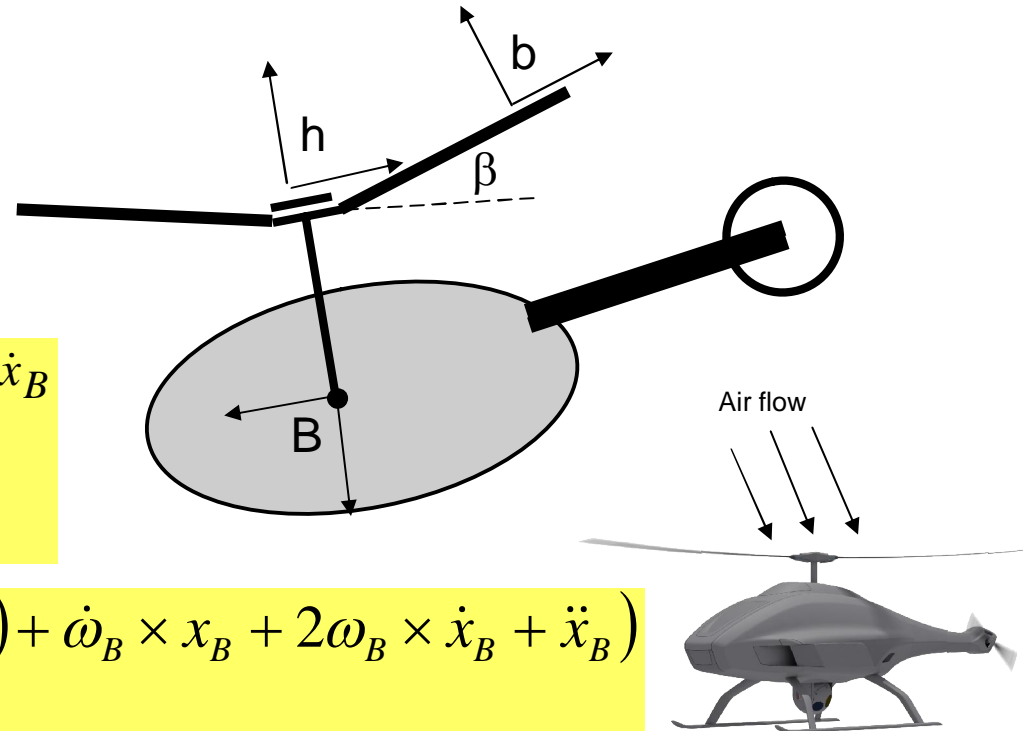
$$\dot{x}_B = A_{h2B} \dot{x}_h$$

$$\dot{x}_h = \dot{x}_{b,0} + A_{b2h} (\omega_b \times x_b) + A_{b2h} \dot{x}_b$$

$$\ddot{x}_E = \ddot{x}_{B,0} + A_{B2E} (\omega_B \times (\omega_B \times x_B) + \dot{\omega}_B \times x_B + 2\omega_B \times \dot{x}_B + \ddot{x}_B)$$

$$\ddot{x}_B = A_{h2B} \ddot{x}_h$$

$$\ddot{x}_h = \ddot{x}_{b,0} + A_{b2h} (\omega_b \times (\omega_b \times x_b) + \dot{\omega}_b \times x_b)$$



E=Earth, B=Body, h=hub and b=blade